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Performance Analysis of a Small-Scale Orthopter-Type Vertical Axis Wind Turbine

by

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Orthopter-type Vertical Axis Wind Turbines (O-VAWT) are energy capturing devices which are suitable for micro-generation in urban areas. This paper aims to study the effect of number of blades (solidity), Aspect Ratio (AR), and wind speed on the performance of a small-scale O-VAWT using experimental and numerical techniques. Wind tunnel experiments were carried out to measure the torque of two-, three, and four-bladed turbine, each at a set of different wind speeds and blade aspect ratios. Increasing solidity was achieved by either decreasing the blade aspect ratio or increasing the number of blades, both resulting in a higher peak power coefficient ($C_{p,max}$) of the VAWT. The startup characteristics of the O-VAWT were also examined at different aspect ratios, and showed its capability to start-up even at low wind speeds. In addition, numerical simulations were carried out using an unsteady three-dimensional Delayed Detached Eddy Simulation (DDES) with Spalart Allmaras model. Numerical results of power coefficient were validated by comparison against available experimental data. Moreover, velocity and vorticity contours for the two-, three-, and four-bladed turbines, and pressure contours at different span-wise locations were thoroughly analyzed to link the flow field aerodynamics to relevant changes in power coefficient for different rotor solidities.

Keywords: Orthopter-type Vertical Axis Wind Turbine, Solidity, Delayed Detached Eddy Simulation

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23 1. Introduction

24 Wind energy has been gaining more popularity as a reliable renewable energy source. Power
25 harvesting has been relying on Horizontal- and Vertical-Axis Wind Turbines. Small-scale designs of the
26 latter have been recently capturing considerable interest due to their capability of harvesting wind energy
27 in all directions at a lower noise levels compared to the HAWTs, making them very suitable candidates
28 for urban areas. In addition, micro-VAWTs are usually placed near obstacles such as buildings and trees,
29 therefore causing unsteadiness in the wind inflow to the turbine. Wekesa et al. [1] conducted a study on
30 the energy content in unsteady winds and concluded that the highest frequency of wind fluctuations
31 suitable for wind turbine applications is 1 Hz. In addition, numerical and experimental studies on the
32 unsteady rotor performance have been investigated by various researchers [2–4].

33 VAWTs are divided into two main categories, Darrieus-type (D-VAWT), a lift based turbine, and
34 Savonius-type (S-VAWT), which is a drag based turbine. Various configurations of Darrieus- and
35 Savonius-type turbines have been recently investigated by researchers in an attempt to improve their
36 performance. Jin et al. [5] conducted a review on basic experimental and numerical research methods
37 employed to assess the performance of Darrieus Vertical Axis Wind Turbines (D-VAWTs). Bhutta et al.
38 [6] discussed the advantages and disadvantages of different VAWT configurations, including various
39 techniques used to optimize their designs.

40 Rezaeiha et al. [7] investigated the effect of pitch angle as a potential way to enhance the performance
41 of a VAWT. Computational Fluid Dynamics (CFD) results suggested that this parameter highly influences
42 the load distribution between upwind and downwind halves of the turbine, thereby making dynamic
43 pitching an encouraging approach to optimize the performance. Solidity is yet another important
44 parameter that influences the turbine performance. However, limited studies were dedicated to
45 investigating its effect on the power coefficient and aerodynamic characteristics of the VAWT. There is
46 no clear evidence that the effect of solidity is similar for different VAWT types. In addition, solidity can
47 be manipulated by altering the blade chord length, the blade number or the rotor radius, with no proof that
48 changing any of these parameters will lead to analogous results.

49 Li et al. [8] performed wind tunnel experiments to understand the effect of solidity at different number
50 of blades on aerodynamic forces around a straight-bladed VAWT. It was concluded that power coefficient
51 decreases when solidity increases, while torque coefficients increase. Similarly, Abu-El-Yazeid et al. [9]
52 investigated the effect of number of blades on the performance of D-VAWT. It was found that the

53 maximum power coefficient is obtained with the lowest number of blades (2 blades). However, decreasing
54 the number of blades contributes to an increase in the radial cyclic aerodynamic forces, a structurally
55 undesirable effect. Another numerical study aiming to reduce the torque variation by manipulating the
56 number of blades on a D-VAWT was conducted by Raciti Castelli et al. [10]. Similar to the previously-
57 mentioned findings by Abu-El-Yazeid et al. [9] and Li et al. [8] an increase in number of blades causes a
58 decrease in the radial component of the aerodynamic forces. Moreover, an increase in solidity is
59 unfavorable since it results in reducing the power coefficient.

60 Previous results on the reduction in power coefficient at higher solidities are not in line with the
61 findings of other researchers. Eboibi et al. [11] conducted an experimental investigation of the influence
62 of solidity on the performance of VAWTs. However, unlike previously mentioned studies, solidity was
63 manipulated by altering the blade chord instead of blade number. Authors claimed that irrespective of the
64 parameter altered, the effect of solidity on VAWT performance is similar. It is worth noting that since the
65 chord length directly affects the Reynolds number, hence, the two designs were tested at different
66 velocities to maintain similar Reynolds numbers. Increasing solidity did not affect the peak power output
67 significantly. Nevertheless, the maximum power coefficient is attained at lower Tip Speed Ratio (TSR)
68 for higher solidities. Cheng et al. [12] studied the dynamics of floating straight-bladed VAWT with a
69 number of blades ranging from two to four. The main advantage that was obtained by increasing the
70 number of blades from 2 to 3 is in the reduction of the tower base bending moment, therefore reducing
71 the fatigue damage. Further improvement by increasing the number of blades to 4 was not observed.
72 Furthermore, increasing the chord length results in an increase in the power coefficient up to a maximum
73 value after which it rapidly decreases. Delafin et al. [13] conducted CFD simulations to understand the
74 effect of number of blades and solidity on the performance of \emptyset -shape VAWT. The number of blades was
75 increased from 2 to 4 while keeping the solidity constant with no observed benefit related to increasing
76 power coefficient. Yet, it is desirable since it results in a decrease in torque, thrust and radial cyclic
77 aerodynamic forces. Subramanian et al. [14] investigated the effect of solidity on the power produced by
78 two and three-bladed small-scale H-type D-VAWTs using three-dimensional CFD simulations. Results
79 showed that the three-bladed configuration having a higher solidity performs better at low tip speed ratios
80 (TSR) due to better interception of blades with wind. On the other hand, vortex-blade interaction at higher
81 TSR reduces the power generation capability of the three-bladed VAWT thereby making lower solidity
82 more favorable.

83 S-VAWTs are drag-driven devices that perform better than D-VAWTs at low wind speeds. Their
84 solidity is usually altered by changing the number of arc-type blades used. Wenehenubun et al. [15]
85 conducted experiments on one, two, three and four-bladed S-VAWTs. It was concluded that the
86 turbine with highest solidity performs best at lower TSR, but the three-bladed configuration
87 outperforms it at higher TSR. On the contrary, experimental results of Mahmoud et al. [16] showed a
88 better performance for the two-bladed configuration at all TSR. Mao et al. [17] investigated the effect
89 of blade arc angle on the performance of S-VAWT, and concluded that arc angle of 160 degrees
90 produced the highest power coefficient for the studied VAWT configuration.

91 Unlike D-VAWT and S-VAWTs, Orthepter-type vertical axis wind turbine (O-VAWT) have received
92 very few consideration. In addition, studies focusing on improving their performance are insufficient.
93 Bayeul-Laine et al. [18] confirmed that this type of turbines performs better than classical VAWTs for
94 some specific blade stagger. Two blade shapes (elliptical and straight) were examined with no clear
95 conclusions about a favorable shape could be drawn since the performance is highly dependent on the
96 TSR range. Cooper and Kennedy [19] tested a symmetric-blade O-VAWT design and obtained a
97 maximum power coefficient of 0.25 with a high startup torque, rendering this design favorable for
98 pumping applications. Elkhoury et al. [20] simulated the flow around an Orthopter-type small-scale
99 VAWT (O-VAWT) with three blades. Effects of wind speed and inflow angle on a three-bladed turbine
100 performance were examined.

101 O-VAWTs possess performance characteristics such as the self-start capability that make them
102 suitable for micro-generation and pumping applications. However, the literature is lacking in studies on
103 O-VAWTs performance. Therefore, the main aim of the present work is to experimentally and numerically
104 investigate the effect of solidity, aspect ratio, and incoming wind speed on the performance of O-VAWTs.
105 The solidity was varied in two different ways: changing the number of blades or changing the chord
106 length. The impact of aspect ratio and wind velocity, previously studied by Elkhoury et al. [20] for a three-
107 bladed O-VAWT, were further extended to include two, and four-bladed turbines. In addition, CFD
108 simulations using DDES were employed to obtain flow patterns of velocity, pressure and vorticity. The
109 analysis of these patterns for different numbers of blades can shed light on changes in flow characteristics
110 due to blade flow interaction.

111

112 2. Experimental Description

113 2.1 Wind tunnel facility

114 The validity of a numerical study should be assessed by comparison of the results against experimental
115 data. A detailed schematic of the wind tunnel and measurement devices is shown in Fig. 1. The
116 experimental setup used in this work consists of an open-circuit wind tunnel with a square test section of
117 $1.25\text{ m} \times 1.25\text{ m}$ and a 2.0 m long working section with a bell mouth of 2.00 m . The wind turbine was
118 installed at the center of cross-section 850 mm downstream from the nozzle exit. The turbulence intensity
119 level and flow non-uniformity in the working section at a wind speed of 8 m/s were less than 0.5% and
120 $\pm 1.0\%$, respectively.

121

122 2.2 Rotor models

123 Fig. 2 shows a photograph and isometric view of the O-VAWT. The O-VAWT tested in the wind
124 tunnel consists of aluminum blades of 1.7 kg each and different thickness of $4, 5$ and 6 mm . A chain-and-
125 sprockets arrangement was used to control the pitch angle of the blade. The blades of turbine rotate with
126 a phase difference about their own axis at half the speed of the main rotor. The Reynolds number based
127 on the blade chord at a wind speed of 10 m/s and the blades dimension are summarized in Table 1. It is
128 important to note that a constant blade weight is maintained by changing the thickness so as to maintain a
129 constant load on bearings and to ensure the same inertial effects on start-up characteristics.

130 **Table 1:** Blade geometry and Reynolds number at different aspect ratios and incoming wind velocities

	Chord length C	Height h	Thickness t	$V = 5\text{ m/s}$	$V = 10\text{ m/s}$
AR=1.0	0.400 m	0.400 m	$4 \times 10^3\text{ m}$	1.4×10^5	2.8×10^5
AR=1.5	0.290 m	0.434 m	$5 \times 10^3\text{ m}$	1.0×10^5	2.0×10^5
AR=2.0	0.228 m	0.455 m	$6 \times 10^3\text{ m}$	0.8×10^5	1.6×10^5

131

132 2.3 Measurement and instrumentation procedures

133 The main measurement devices consist of an ultrasonic anemometer (Kaijo Sonic, DA-650-3TH &
134 TR-90AH) that is used to measure the speed and temperature of the incoming air flow. The averaged
135 torque T of the turbine was measured by a torque transducer (TEAC TQ-AR5N with a rate capacity of 10
136 N.m) along with a digital tachometer (ONO SOKKI HT-5500) for measuring the rotational speed, rpm ,

137 of the turbine. The output signal of the torque transducer was converted by a 16-bit A/D converter with a
138 sampling intervals of 0.5 degrees, and 36,000 items (50 revolutions) of data were stored. The power
139 coefficient of the turbine was measured at different TSR for constant wind speeds of $V = 5$ and 10 m/s, by
140 varying the rotational speed of the geared motor (Mitsubishi Electric, GM-S 2.2 kW) accompanied by a
141 variable-frequency inverter (Hitachi, SJ200). The measured values correspond to the total torque including
142 all mechanical losses of bearings and gears. To account for these losses however, the following equation
143 was added to the results of the numerical simulations $C_{p,loss} = 0.0005 - 0.0355\lambda - 0.0502\lambda^2 - 0.3124\lambda^3$
144 . This equation was obtained using a curve fit of the experimental data with a coefficient of determination
145 R^2 of 0.9999. The uncertainty levels of measurements were estimated to around 2% for the ultrasonic
146 anemometer, ± 0.03 N. m for the torque transducer, and around 1% for the tachometer, respectively. The
147 uncertainty of maximum power coefficient at $V = 10$ m/s was approximately estimated to be $\pm 0.05C_p$.

148

149 **3. Numerical Model**

150 **3.1 Turbulence Model**

151 CFD simulations were performed by employing the DDES based on the Spalart Allmaras (SA)
152 turbulence model. DDES falls in the category of hybrid Reynolds-Averaged-Navier-Stokes/Large-Eddy-
153 Simulation (RANS/LES) models, which is capable of achieving higher accuracy than RANS models for
154 complex flows with less computational cost when compared to LES. The choice of the DDES based on
155 the SA model stems from the fact that it was successful in modeling aeronautical applications involving
156 massive flow separation [21], not to mention the lower computational cost associated with solving a single
157 transport equation. In addition, the model has a linear near-wall behavior of its transport property,
158 requiring mesh densities as low as those used with algebraic models [22].

159

160 **3.2 Computational Domain**

161 **3.2.1 Geometry and Mesh**

162 To account for complex flow structures associated with various aspect ratios, a 3-D approach is
163 considered in the present study, which allows to model the central shaft along with the connecting rods. It
164 is worth noting that all modeled turbines are simpler than those of the experiment, where gears and chains
165 are omitted for the purpose of improving the mesh quality and solver's convergence time with minimal

166 effect on the results. The different turbines shown in Fig. 2 have similar structure, where each blade is
 167 attached to the rotor by two connecting rods, and the latter are connected to the blade by the means of two
 168 blade holders. Thus, each blade is allowed to rotate around its center with an angular velocity of $-\omega/2$
 169 relative to its rotation around the rotor as indicated in Fig. 2. As depicted in Tables 2 and 3, ten turbines
 170 are analyzed, each of the two- and three-bladed VAWTs are simulated with three different aspect ratios
 171 of 1, 1.5 and 2 while the four-bladed one is only simulated with the aspect ratios 1.5 and 2 since
 172 interference/intervention among blades occurs at AR=1. For the first eight turbines shown in Table 2, the
 173 aspect ratio was altered by changing both the blade chord and height while maintaining constant blade
 174 swept area as defined in Fig. 2. Moreover, two additional turbines with constant height and varying blade
 175 chord as shown in Table 3 were tested. The diameter of all considered rotors is 51 cm. The 3-D
 176 investigation of such turbines requires a segmentation of the domain into the far field, turbine and blades
 177 domains as depicted in Fig. 3. Each blade domain comprises of a rotating cylinder that is encapsulated by
 178 the turbine's domain being a rotating domain as well. The full assembly is engulfed by the far field that
 179 represents the fluid flowing around the VAWT.

180 **Table 2: Solidities and swept areas for turbines with different aspect ratios and number of blades**

	N=2	N=3	N=4
AR=1.0 (C=0.4m, H=0.4m)	$\sigma = 0.50$ A=0.284	$\sigma = 0.749$ A=0.284	----
AR=1.5 (C=0.29m, H=0.435m)	$\sigma = 0.362$ A=0.284	$\sigma = 0.543$ A=0.284	$\sigma = 0.724$ A=0.284
AR=2.0 (C=0.228m, H=0.456m)	$\sigma = 0.285$ A=0.284	$\sigma = 0.427$ A=0.284	$\sigma = 0.570$ A=0.284

186 **Table 3: Solidities and swept areas for turbines with different blade chord lengths**

	N=3
AR=0.89 (C=0.449m, H=0.4m)	$\sigma = 0.841$ A=0.294
AR=1 (C=0.4m, H=0.4m)	$\sigma = 0.749$ A=0.284
AR=1.06(C=0.377m, H=0.4m)	$\sigma = 0.706$ A=0.279

187

188

189

190 **3.2.2 Far Field**

191 As stated earlier, this domain is equivalent to the wind tunnel where the fluid flows across from
192 one end to another. It is worth noting that the dimensions are carefully selected in order to prevent any
193 interference that may lead to inaccurate results based on the recommendations of Rezaeiha et al. [23].
194 Therefore, the turbine is placed at a distance of 11.37 D from the inlet and 12.13 D from the outlet, and
195 midway from the other boundaries in the transverse and rotor axial direction as depicted in Fig. 3. A
196 Boolean operation takes place to subtract the rotating domains from the fixed one to create separate
197 entities. Thus, an interface boundary condition is assigned at the intersection of the two domains so the
198 turbine can rotate around its center while allowing a smooth flow between the two. An unstructured coarse
199 mesh is applied since the geometry is of simple shape and flow characteristics in this region are not
200 supposed to change. Moreover, identical sizing functions are enforced at the interfaces to ensure a smooth
201 transmission between the two without incurring any loss due to interpolation of data.

202

203 **3.2.3 Rotor Domain**

204 In addition to being a rotating domain, this field encompasses the full geometry, such as the blades,
205 connecting rods, and rotor. For simplicity purposes, the turbine excluding the blades is removed from this
206 area leaving thin surfaces with no slip boundary condition, in return this reduces the number of elements
207 and accelerates the solver's convergence. Similarly, the blade domains are deducted from the rotating
208 volume while an interface is set at the intersection to ensure continuity of the flow between the different
209 domains and allow each domain to rotate together and relative to each other. Hence, when an angular
210 velocity ω around the z-axis is allocated to the domain, both the turbine and blade domains follow the
211 same path at the same rate. Additionally, a fine unstructured mesh is applied given the complexity of the
212 geometry. Identical sizing function is also imposed at the interfaces of the rotating cylinders to obtain a
213 good transition between the domains without jeopardizing the accuracy of the results.

214

215 **3.2.4 Blade Domain**

216 As mentioned earlier, each blade is enclosed by a cylinder that is separated from the turbine domain
217 since this VAWT is characterized by their variable pitch mechanism. The blades are deducted from their

218 enclosures and removed, consequentially a thin surface with non-slip boundary condition remains
 219 covering the boundaries of all blades. This in return, reduces the mesh density and helps in accelerating
 220 the solver’s convergence. This set-up allows for the blade to rotate around its mid center with an angular
 221 velocity of $-\omega/2$, and around the turbine’s center with an angular velocity of ω , as stated previously. Seeing
 222 that the blades are the most important parts of a VAWT, The finest unstructured mesh is applied to the
 223 blades at this level along with inflation layers adjacent to their surfaces as they comprise the most
 224 important part of a VAWT. Hence, complex flow structures near the blade’s wall can be accurately
 225 resolved, which in turn increases the reliability of the results.

226

227 **4 Validation**

228 **4.1 Mesh Dependency Study**

229 Mesh dependency study is crucial for insuring accurate results. Therefore, simulations on the four-
 230 bladed turbine with $AR = 2$, $V = 5\text{m/s}$ and $TSR = 0.40$ were carried out using different mesh resolutions.
 231 After convergence was obtained for an initial mesh, referred to herein as “Regular mesh”, the grid was
 232 further refined in the turbine and blade domains and iterations were performed until solution converged
 233 to the set criterion of 1×10^{-4} of scaled residual. Specifications of both grids are delineated in Table 4. The
 234 criterion used to confirm that the solution is mesh independent is the change in the averaged power
 235 coefficient C_p . As depicted in Fig. 4, the variation in C_p between regular and fine mesh solutions is less
 236 than 2% in averaged power coefficient. Therefore, any further refinement of the mesh is not justified in
 237 terms of accuracy, and has a disadvantage of increasing the computational cost. It is worth to note that 10
 238 to 15 layers of inflation prisms were placed around all solid surfaces with the first off-surface node located
 239 within a range of $1 \times$ to 5×10^{-6} m, resulting in values of $y^+ \leq 1.0$.

240 **Table 4: Regular and fine grid specifications for mesh dependency study**

Domain	Regular Mesh	Fine Mesh
Blade 1 CV	1,610,660	3,785,943
Blade 2 CV	1,614,903	3,808,216
Blade 3 CV	1,588,363	3,768,389
Blade 4 CV	1,614,661	3,799,238
Turbine CV	2,127,497	3,879,281
Far Field	441,196	441,196
Total Cell Number	8,997,280	19,482,263

241

242 4.2 Time Step Study

243 For the same geometry and regular grid density of the preceding section, the initial time step used was
244 $\Delta t = 5 \times 10^{-5}$ s corresponding to the time needed for the turbine to accomplish a rotation angle of 0.045
245 degrees at TSR = 0.4, and resulting in a CFL ≤ 1.0 adjacent to the rotor and in wake regions. This condition
246 was maintained throughout the computations to ensure proper scale-resolving simulations. This time step
247 was further reduced to half of this initial value. Fig. 5 depicts the variation of power coefficient for both
248 time steps. The maximum percentage difference in C_p is less than 0.56%. Hence, the larger time step was
249 adopted since it minimizes the computational cost while retaining good accuracy levels. The averaging
250 process of C_p values started after the elapse of the initial two revolutions and continued over the following
251 six to eight revolutions, which were enough to obtain statistical convergence [24].

252

253 4.3 Flow Solver and Boundary Conditions

254 A commercial CFD solver, Ansys® Fluent® 17.2, was utilized to solve the equations of motion. An
255 unsteady coupled pressure-based double precision solver was employed. A second order bounded central
256 difference discretization scheme was adopted for the momentum equations whereas a second order
257 upwind-based discretization scheme was selected for flow variables of the SA model. Numerical
258 instabilities resulting from the use of central difference scheme were avoided by adopting the bounded
259 central difference scheme. The latter is more robust compared to the former due to its slight numerical
260 dissipation however, is sufficiently low to allow turbulent structures to evolve. Flow gradients were
261 discretized using the Least Square Cell Based approach. A second order central difference pressure
262 interpolation scheme, which reconstructs the pressure values at cell faces, was selected. A second order
263 implicit scheme was employed for transient formulation. Default under-relaxation factors of 1.0 for
264 density, body force, and turbulent viscosity, and 0.8 for the modified turbulent viscosity were employed.
265 The default explicit-relaxation factor of 0.75 was adopted for momentum and pressure. The solver was
266 utilized in a parallel mode with 14 processors, 112 cores of Intel®-Xeon®-CPU-E5-2667-v3-@-3.2GHz.
267 A typical revolution on these cores consumed around 23 hrs of computational time on the regular mesh.

268 All considered test cases fall within the incompressible flow regime and thus, the following boundary
269 conditions were employed. A velocity inlet boundary condition was implemented at the inlet where a
270 uniform inflow velocity was assumed. The inlet pressure was extrapolated upstream from the computation

271 zone. At the outlet, the pressure outlet boundary condition was implemented where an outlet static gauge
272 pressure of zero was specified. All other flow variables were extrapolated from interior cells. A symmetry
273 boundary condition was implemented in the transverse and axial directions as indicated by Fig. 3.
274 Gradients of all flow variables are set to zero on these boundaries, thereby neglecting blockage effect.
275 Based on the work of Rezaeiha et al. [25], it was found that a blockage ratio of less than 5% had negligible
276 effect on the predicted power performance of the turbine. It is worth to note that a blockage ratio of 1.29%
277 based on the swept area of the rotor, A , and the domain cross sectional area ($5D \times 17D$) was obtained for
278 the current configuration, justifying the use of a symmetry boundary condition. The no-slip and condition
279 was applied at all solid boundaries.

280

281 **5 Results**

282 **5.1 Effect of Chord and Number of Blades (Solidity) on Performance**

283 Solidity is one of the main parameters that affect the power produced by the turbine. Manipulating
284 solidity is usually achieved by varying the number of blades or the blade chord length. Different
285 experimental approaches could be used to measure the power coefficient, the torque transducer was
286 employed in this work. Fig. 6 compares the averaged power coefficient variation with TSR for VAWTs
287 with different number of blades, but similar aspect ratios, at two different wind speeds. The trend in C_p -
288 TSR curves is similar to the experimental results of the D- and S-VAWTs as presented by Li et al. [8],
289 Eboibi et al. [11], and the numerical results of Consul et al. [26]. The power coefficient increases with
290 TSR, until it attains a peak value, after which it begins to decrease. In the region past the peak TSR value,
291 the rate at which C_p decreases is affected by solidity. Higher solidities result in faster decrease in C_p values.
292 Fig. 6 results also show a shift in the C_p -TSR curves towards lower TSR values when the number of blades
293 increases. This is further confirmed in Fig. 7 where the TSR at which maximum C_p occurs is plotted for
294 all tested VAWT models. Adding more blades to the VAWT results in reaching the maximum C_p at lower
295 TSR. In addition, the operating range of the VAWT increases at lower solidities. This finding is in
296 agreement with the results of Eboibi et al. [11] and Consul et al. [26] which showed that one of the
297 consequences of lowering solidity of a D-VAWT is a wider range of the C_p -TSR curve. The effect of
298 solidity on the maximum output power of the turbine could be inferred from Fig. 8. It is clear that the peak
299 power coefficient produced increases with the number of blades. Increasing the number of blades from 2
300 to 3 results in a 10% increase in peak power coefficient at $V = 5\text{m/s}$, and 35% at $V = 10\text{ m/s}$. Moreover,

301 the four-bladed architecture produces a higher C_p compared to the three-bladed one at both tested wind
302 velocities. On the contrary, increasing the VAWT solidity by solely increasing the blade chord length does
303 not lead to similar results. As shown in Fig. 9, the VAWT with highest solidity produces more power at
304 all TSR. However, the change in solidity did not cause any shift in the peak of C_p -TSR curve towards
305 lower TSR as observed when manipulating the number of blades. However, it is worth noting that the
306 range of maximum increase in solidity when changing c from 378 to 445 mm is 17%, which is much lower
307 than the 50% increase obtained when changing the number of blades from 2 to 3.

308 In order to assess the ability of the DDES model to predict the performance of the VAWT, CFD
309 simulation were performed at a velocity of 10 m/s for all turbines, the results of which are shown in Fig.
310 6 (d, e and f). The maximum deviation from experimental data occurs at or slightly off the peak power
311 coefficient. Discrepancies between experimental and numerical results could be associated with the
312 simplifications in the modeled geometry and/or accuracy of selected model, a point that merits further
313 investigation.

314

315 **5.2 Effect of Aspect Ratio and Wind Speed on Performance**

316 In the previous section, results regarding the performance of O-VAWTs with different blade chord
317 length were discussed to infer the effect of solidity. Additionally, changing chord length also leads to a
318 change in aspect ratio. Nevertheless, the range of aspect ratios that could be achieved without changing
319 the blade height is short, and consequently the obtained difference in C_p was within the uncertainty of the
320 measurements weakening the judgement regarding the effect of aspect ratio on the power produced by the
321 turbine. Therefore, additional experiments were carried out for different combinations of c and h , so that
322 a wider range of aspect ratios is studied and shown in Fig. 6. Combinations of c and h were chosen such
323 that the turbine swept area (A) is conserved for all tested VAWT models. By examining the results shown
324 in Fig. 7a and b, it can be concluded that the impact of aspect ratio on the speed at which $C_{p,max}$ occurs
325 does not show clear trends. For the two- and four-bladed designs, increasing the aspect ratio did not affect
326 the location of the maximum C_p at both wind velocities. Furthermore, increasing the aspect ratio while
327 keeping the same number of blades lowers the peak power coefficient as shown in Fig. 8. Such behavior
328 was observed by Zanforlin et al. [27] for small straight-bladed VAWT at low Reynolds number and was
329 attributed to the increase in Reynolds number implied by the chord extension. Therefore, at higher wind
330 speeds ($V=10$ m/s), the favorable effect of increasing the Reynolds number is counteracted by the

331 detrimental growth in losses due to tip vortices, and the growth in C_p at lower AR is less important.
332 Nevertheless, this behavior is in contradiction with the findings of Abulyazied et al. [9] and Raciti Castelli
333 et al. [10] for D-VAWTs. Consequently, increasing solidity by either increasing number of blades or
334 decreasing aspect ratio (chord) is favorable for increasing the power produced by the VAWT.

335 It can also be inferred by comparing Fig. 6 (a and d, b and e, c and f) that the power coefficient
336 increases with wind speed. Additionally, the range in which positive values of C_p are maintained is much
337 wider at higher wind velocities. It is also worth noting that the difference in power coefficients at wind
338 speed of 5 m/s between the two- and three-bladed designs is not significant. However, the four-bladed
339 design shows an average increase of 40% in $C_{p,max}$. On the contrary, the increase in C_p at a wind speed of
340 10 m/s by increasing the number of blades from 2 to 3 is substantial (around 35% increase in $C_{p,max}$), while
341 it is less significant for a further increase in number of blades from 3 to 4. The TSR corresponding to a
342 maximum C_p increases with wind speed, which is in agreement with the results of Elkhoury et al. [20] for
343 a three-bladed O-VAWT.

344

345 **5.3 Effect of Solidity on Turbine Start-Up Characteristics**

346 Wind turbines could be used as standalone power generation systems or in hybrid mode with other
347 renewable or non-renewable energy sources. When employed to supply electricity in remote areas, the
348 ability of the wind turbine to self-start at low wind speeds is crucial. D-VAWTs have low starting torque
349 and therefore are not suitable for such applications. On the other hand, S-VAWTs have good self-starting
350 characteristics, but fail to achieve high power coefficients. In Fig. 10, the power coefficient of the O-
351 VAWT is compared to that of a S-VAWT tested by Blackwell [28], both having an aspect ratio of 1 at an
352 incoming wind velocity of 7 m/s. The O-VAWT is able to achieve a higher peak C_p than the S-VAWT
353 and at lower TSR. In addition, Elkhoury et al. [20] showed that the O-VAWT has the ability to self-start
354 at wind speeds as low as 1.3 m/s and thus, outperforming the S-VAWT which self-starts at 2 to 6 m/s wind
355 speeds.

356 Solidity has a strong impact on the self-starting characteristics of the VAWT. As pointed out in the
357 previous section, solidity could be varied by changing either the number of blades or the blade chord
358 length. Elkhoury et al. [20] thoroughly investigated the effect aspect ratio manipulated by changing both
359 the blade chord and height. It was concluded that the O-VAWT with AR=1 not only starts at lowest wind

360 speeds, but also attains high TSR at lower wind speeds when compared to the turbines with AR equal 1.5
361 and 2. In Fig. 11, the effect of changing the blade chord while retaining same height on the turbine self-
362 start wind speed is shown. Similar superior self-starting behavior of the VAWT with lowest aspect ratio
363 (highest chord and solidity) is observed. Fig. 12 shows the effect of number of blades on the turbine startup
364 characteristics. The three-bladed configuration is able to start at a wind speed less than 2 m/s, thereby
365 outperforming the two- and four-bladed VAWTs. However, the turbine with lowest solidity (N=2) has a
366 sharper increase in TSR making it able to achieve higher tip speed ratios than three- and four-bladed
367 turbines for wind speeds greater than 2 m/s.

368 Hybrid D/S-VAWT configurations are the subject of many recent studies [29–31]. The main aim of
369 this configuration is to take benefit of the high C_p of D-VAWT and the good self-starting characteristics
370 of the S-VAWT. Fig. 13 shows the power coefficient variation with TSR of a three-bladed O-VAWTs
371 with AR=1 and a three-bladed D-VAWT previously studied by Elkhoury et al. [32]. A hybrid O/D-VAWT
372 configuration could have promising performance characteristics in terms of startup torque and peak C_p .
373 However, further investigations are required to assess the viability of such configurations.

374 Moreover, it is worth noting that the current study results were generated assuming uniform incoming
375 air flow, while in real conditions, the incoming flow is turbulent and unsteady. Wekesa et al. [2]
376 investigated the aerodynamic performance of a small-scale S-VAWT operating in an external turbulent
377 inflow. When a turbulent inflow was introduced, an improvement in the startup characteristics of the
378 turbine was observed and was attributed to the reduction of the relative velocity between the rotor and the
379 incoming wind. However, further investigation on the performance of the O-VAWT under unsteady wind
380 conditions is required to confirm its superior self-starting characteristics in real life applications.

381

382 **5.4 Blade Flow Interaction**

383 In order to explain the observed differences in performance of O-VAWTs with different number of
384 blades, flow patterns of velocity and vorticity were closely examined at an aspect ratio of 1.5 and a wind
385 velocity of 10 m/s. The patterns were obtained from numerical simulations which were previously
386 validated by comparing average C_p values at different tip speed ratios to experimental results. Fig. 14
387 shows the instantaneous out-of-plane vorticity contours on a mid-plane cutting through the turbine blades
388 for 30 degrees blade rotation increments. The flow around blades oriented at 0, 30, and 60 degrees does

389 not interact with the flow features produced by neighboring blades. Therefore, for the two-, three- and
390 four-bladed turbine, the instantaneous contours of vorticity are similar. At 90 degrees, a Leading Edge
391 Vortex (LEV) develops for all turbines. However, it is smallest for the turbine with highest solidity.
392 Moreover, the flow patterns around the 180 degree oriented blades in Fig. 14 (a, b and c) indicate higher
393 concentration of positive and negative vorticity patches in the absence of neighboring blades. When
394 neighboring blades are present as in the case of the three-bladed and four-bladed turbines, the flow
395 becomes more constrained and therefore small vorticity patches tend to combine into a bigger patch on
396 the trailing edge of the blade. Therefore, the energy extraction capability of the turbine is enhanced by
397 adding more blades. When linking this result to the performance of the turbine, the higher power
398 coefficient for the higher solidity turbines can be explained. The formation of rollup vortex with negative
399 vorticity is observed at 270 degrees for the two-bladed turbine in Fig. 14j and at 210 degrees for the three
400 and four-bladed turbines shown in Fig. 14 (k and l). The rollup vortex continues to grow until the azimuthal
401 angle becomes 300 degrees. At this point, the layer of vortices from the neighboring blades interacts with
402 the rollup vortex and results in breaking it up into several smaller vorticity patches. Differences in flow
403 patterns between the two and three-bladed VAWTs are very prominent, while the three and four-bladed
404 turbines have similar flow features. It is worth noting that the discrepancy in C_p between the two turbines
405 of higher solidities is low, while the lowest solidity turbine produces much lower power coefficients than
406 both. Therefore, observations on instantaneous flow patterns are in perfect agreement with average power
407 coefficient results.

408 Mid-plane contours of velocity magnitude for the same configuration are depicted in Fig. 15. An
409 examination of these contours around blades oriented at different azimuthal angles reveals interesting
410 aspects of the flow. The flow around 0 and 30 degree oriented blades for the range of solidities considered
411 is indistinguishable. At 60 degrees (Fig. 15b, g and i), the appearance of a region of high velocity at the
412 blade leading edge indicates the formation of the leading edge vortex described earlier. The fastest growth
413 of the LEV is observed on the lower solidity turbine due to the absence of close neighboring blades. At
414 210 degrees (Fig. 15f), the wake region from the trailing edge of the blade interacts with the LEV and
415 causes its detachment from the blade surface. In addition, the wake regions on blades oriented at 180
416 degrees and higher cover the entire blade surface for the three- and four-bladed O-VAWTs. Therefore, the
417 drag force is higher for turbines of higher solidities. In general, O-VAWTs can be designed to be lift or
418 drag-driven depending on the blade cross-sectional shape. The configuration tested in the present study
419 has a flat cross-section and is therefore drag-driven. In addition, the presence of big regions of low velocity

420 for the higher solidity turbines confirms the attainment of higher C_p as they extract the flow energy better
421 than the two-bladed configuration.

422

423 **5.5 Three-Dimensional Flow Characteristics**

424 In order to better understand the effects of blade aspect ratio and number of blades on the performance
425 of the turbine, three-dimensional flow characteristics were closely examined at an incoming wind speed
426 of 10 m/s and a TSR of 0.4. Since the flow features on each blade change as it rotates, a careful selection
427 of the azimuthal angle of the blade, on which flow characteristics will be examined, is required. Fig. 16
428 (a, b and c) shows Iso-surfaces of pressure at 70 Pa colored by velocity magnitude in the flow direction
429 and pressure contours at different span-wise locations on the blade for the two-bladed turbine with AR of
430 1. A LEV develops on the blade oriented at 60 degrees as shown in Fig. 16a. As the blade rotates to
431 achieve an azimuthal angle of 120 degrees, the LEV is entrained downstream and tip vortices appear on
432 the upper and lower edges of the blade. However, the effect of tip vortices is restricted towards the blade
433 edges ($\mu = 0.98$). It is also worth noting that the LEV remains attached to the blade surface at all span-
434 wise locations. At 180 degrees, the tip vortices are elongated, and about to be shed into the freestream.
435 Moreover, their effect stretches from blade edge to the midspan location causing a separation of the LEV
436 from the blade surface at $\mu = 0$. Since it is interesting to investigate flow features along the blade when
437 vortices are prominent and separation phenomenon occurs, the azimuth angle of 180 degrees was selected
438 to examine the impact of AR and blade number on flow features.

439

440 Fig. 17 (a, b and c) also shows Iso-surfaces of pressure at 70 Pa colored by velocity magnitude in the
441 flow direction and pressure contours at different span-wise locations on the blade for the two-bladed
442 turbine having AR equal 1, AR equal 2, and the three-bladed turbine having AR equal 1 respectively. For
443 all three VAWTs, tip vortices are generated due to the pressure difference between the pressure side and
444 suction side of the blade causing the flow to curl along the blade tip towards the suction side. However,
445 by comparing Fig. 17 (a and b), it is evident that tip vortices are more prominent on shorter blades having
446 low AR. Additionally, vortices on the high AR blade seem to be fed by the large gap between the blade
447 edge and the turbine shaft, while vortices on the low AR blade strongest near the leading edge of the blade.
448 The impact of higher Reynolds number associated with shorter blades counteracts the high tip losses and
449 therefore lower aspect ratios are favorable for the small-scale O-VAWT tested in this work. In addition,

450 pressure contours confirm that tip losses are lowest at midspan ($\mu = 0$) where highest C_p is attained [33],
451 therefore highlighting the negative effect of the three-dimensional nature of the flow on the VAWT
452 performance. Another comparison between Fig. 17 (a and c) could be made to investigate the impact of
453 number of blades on the turbine performance. The addition of a third blade to the turbine results in more
454 concentrated vortices, without causing substantial changes in the flow structures. Nevertheless, the
455 additional blade is favorable for extracting more power from the incoming wind. However, careful
456 investigations are required to confirm the effect of adding extra blades. Similar conclusions could be
457 drawn by examining Fig. 18 (a, b and c). The streamlines along blades having same azimuthal angle reveal
458 a stronger rollup of vortices associated lower AR blade, and no noticeable impact of adding a third blade
459 on the flow characteristics.

460

461 **6 Conclusions**

462 O-VAWTs with number of blades ranging from 2 to 4, and aspect ratios of 0.899, 1, 1.06, 1.5 and 2,
463 were tested in a wind tunnel at wind speeds of 5 and 10 m/s. Additionally, DDES using Spalart Allmaras
464 model were performed at the latter wind speed. Following a close agreement between CFD and
465 experimental results, the effect of solidity, aspect ratio and wind speed on the turbine performance was
466 examined. The major findings are summarized below:

- 467 1- The maximum power coefficient attained by the turbine increases with solidity. However, the
468 range of TSR for which the turbine maintains positive C_p is wider for turbines of lower
469 solidities. In addition, the tip speed ratio at which $C_{p,max}$ occurs decreases for O-VAWTs with
470 higher number of blades (solidity), but does not show consistent behavior for increased solidity
471 by means of chord length.
- 472 2- An increase in aspect ratio, while retaining the same number of blades, results in a decrease in
473 $C_{p,max}$. Nevertheless, no clear trend on the variation of the tip speed ratio at which $C_{p,max}$ occurs
474 with aspect ratio could be inferred from the results.
- 475 3- The TSR corresponding to $C_{p,max}$ increases at higher wind speeds.
- 476 4- The O-VAWT is able to start-up at lower wind speeds compared to a typical S-VAWT.
477 Moreover, decreasing solidity by increasing aspect ratio is favorable for improving the start-
478 up at low wind speeds.

- 479 5- For an incoming wind speed of 10 m/s, differences in instantaneous flow structures between
480 the two- and three- bladed turbines were substantial and therefore, leading to a significant
481 change in the average power coefficient produced. On the other hand, the flow around three-
482 and four- bladed turbines showed similar features, resulting in smaller change in the average
483 C_p at all tip speed ratios.
- 484 6- Three-dimensional flow characteristics reveal that the effect of tip losses on reducing the power
485 coefficient of the turbine having lower AR is counteracted by the increase in Reynolds number
486 due to chord extension.

487 The present work on O-VAWTs sheds light on the effect of different parameters on its performance
488 under uniform inflows. Although turbulent inflows usually enhances the output power of the turbine, it is
489 recommended to conduct further studies to confirm its ability to withstand the increase in aerodynamic
490 loads under such conditions. In addition, hybrid configurations of O/D-VAWTs are believed to possess
491 the self-starting ability of O-VAWT and the energy capturing capabilities of the D-VAWT at high wind
492 speeds. Therefore, future work will be extended to developing and testing such wind turbine
493 configurations.

494

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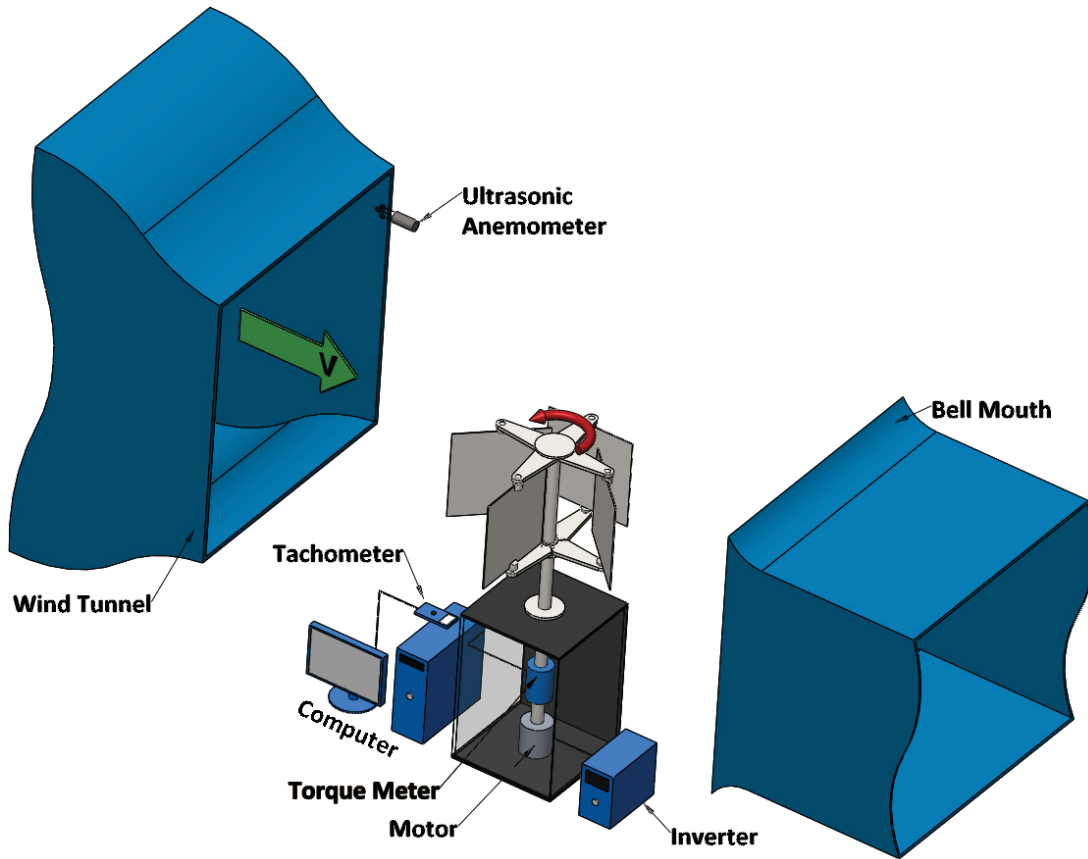
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631 **Nomenclature**

AR	: Aspect ratio of blade ($= h/c$)
A	: Turbine swept area ($= [D + c/2] h$)
C_p	: Turbine power coefficient ($= T\omega / 0.5\rho V^3 A$)
$C_{p,max}$: Maximum pressure coefficient
C	: Blade chord length
D	: Turbine diameter
h	: Blade span length
I	: Turbulence intensity ($= u'/V$)
l	: Turbulent length scale
N	: Number of blades
R	: Turbine radius
Re	: Reynolds number
T	: Turbine torque
TSR	: Tip speed ratio ($\lambda = R\omega / V$)
V	: Wind speed
y^+	: Dimensionless wall distance
ρ	: Air density
σ	: Turbine solidity ($= Nc/2\pi R$)
ω	: Turbine angular velocity ($= 2\pi rpm/60$)
μ	: Non-dimensional span-wise position

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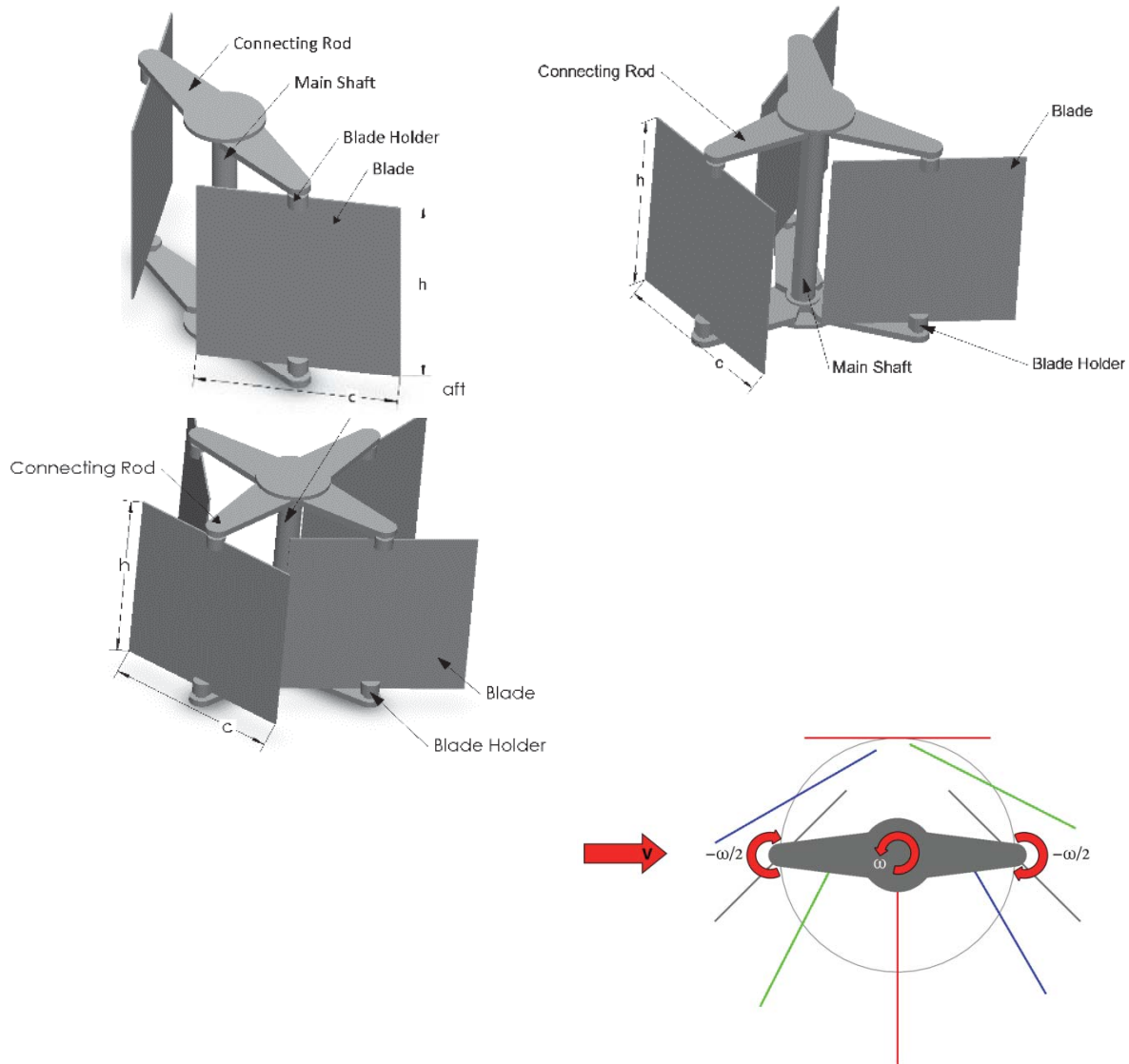


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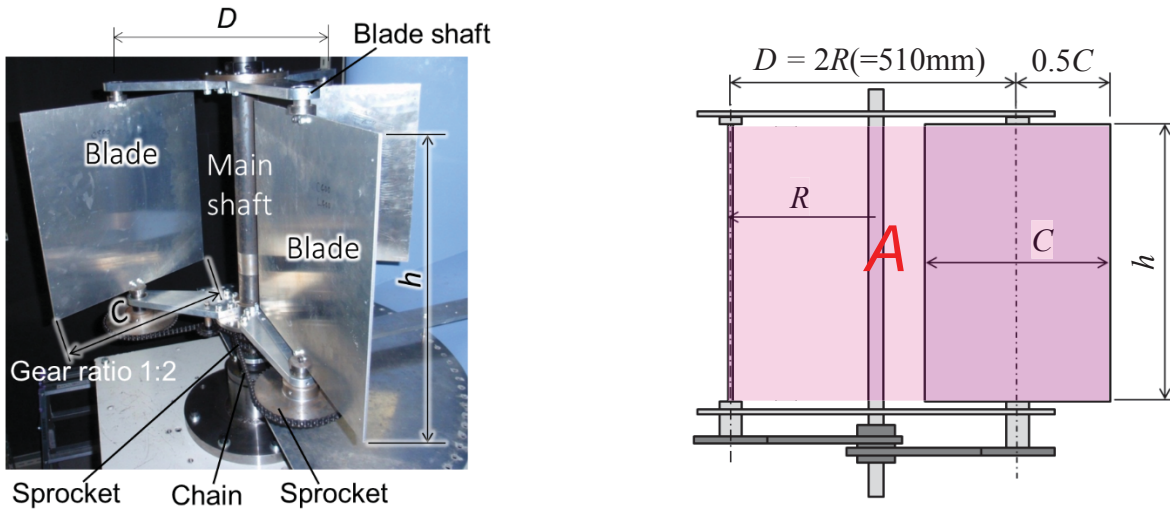
635 **Figure 1: Schematic diagram of the wind tunnel and the experimental apparatus.**

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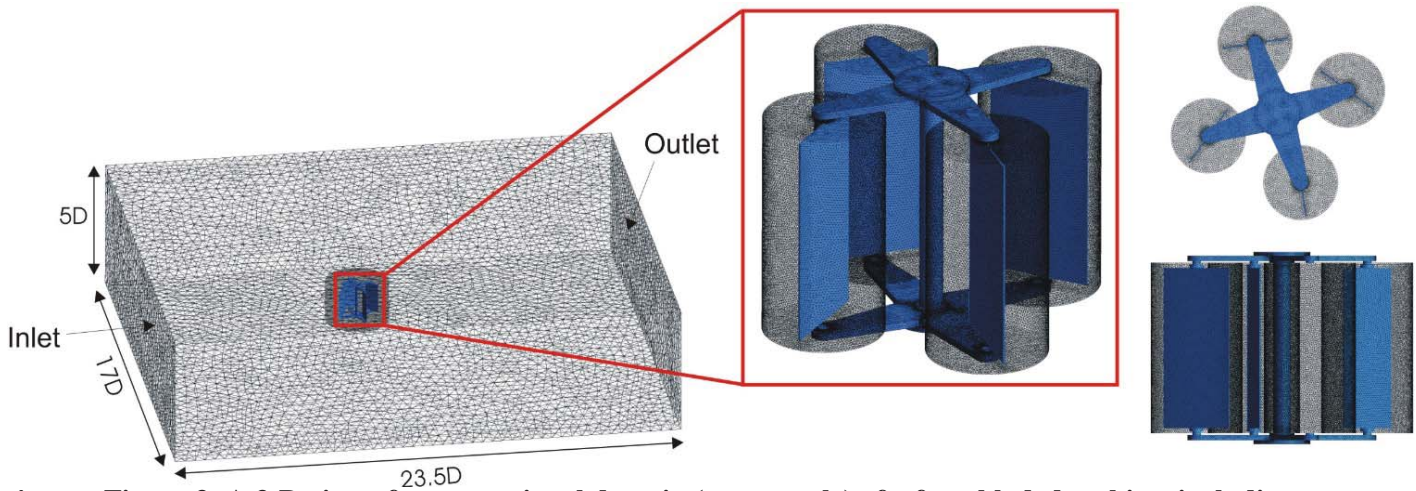


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640 **Figure 2: Isometric view of the three rotors and top view of the two-bladed rotor showing blade motion.**

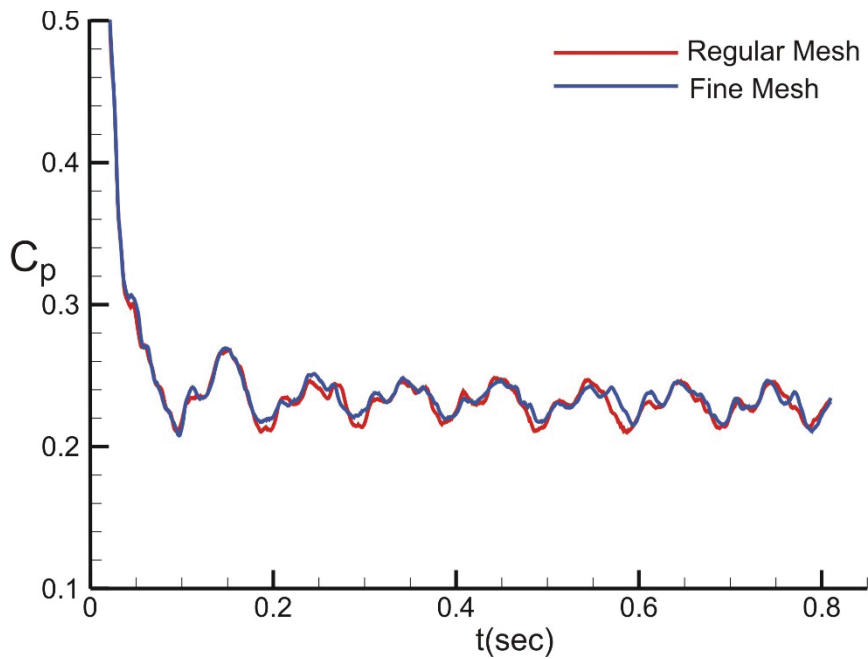
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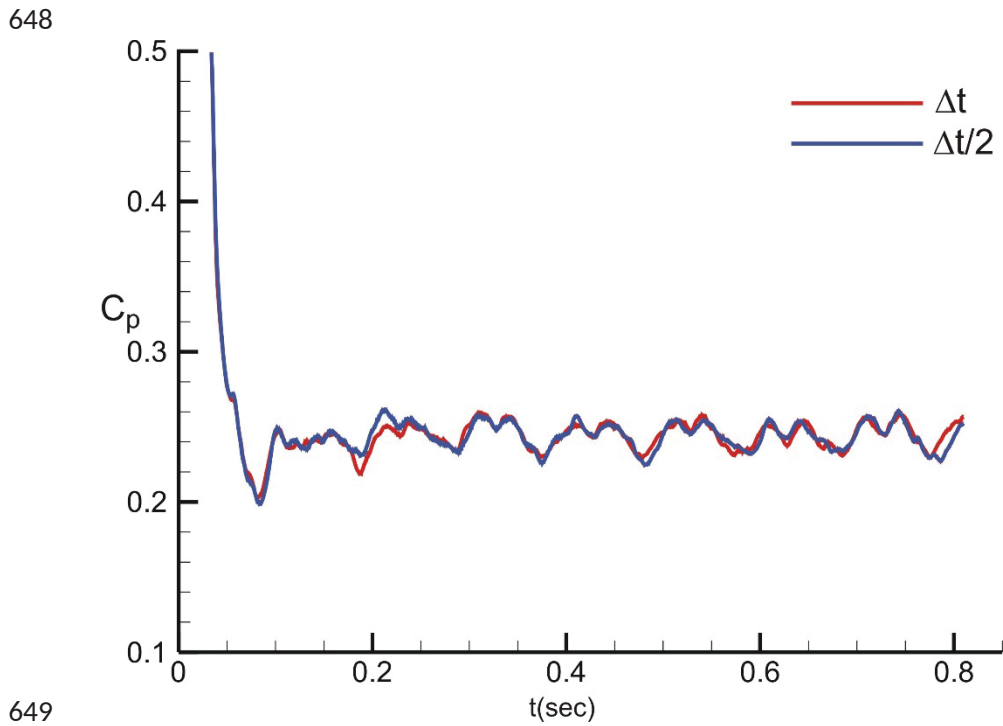


643 **Figure 3: A 3-D view of computational domain (not to scale) of a four-bladed turbine, including an**
644 **isometric, top, and side view of rotor.**

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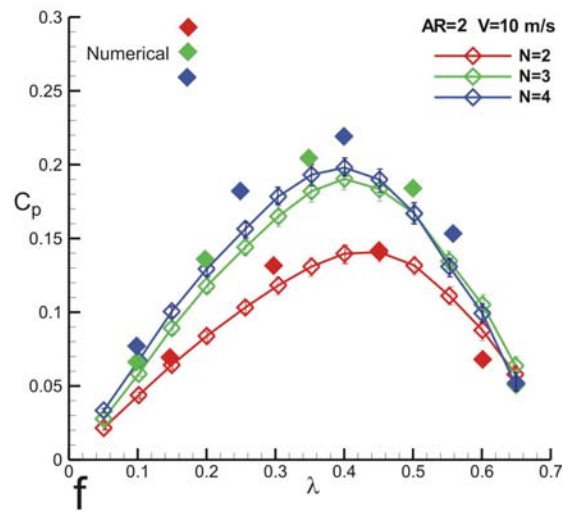
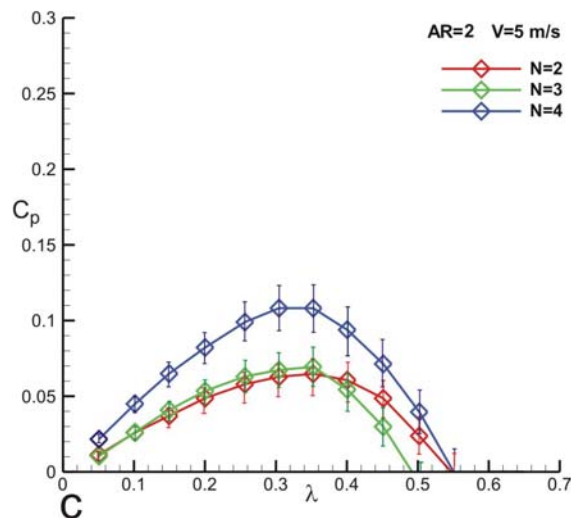
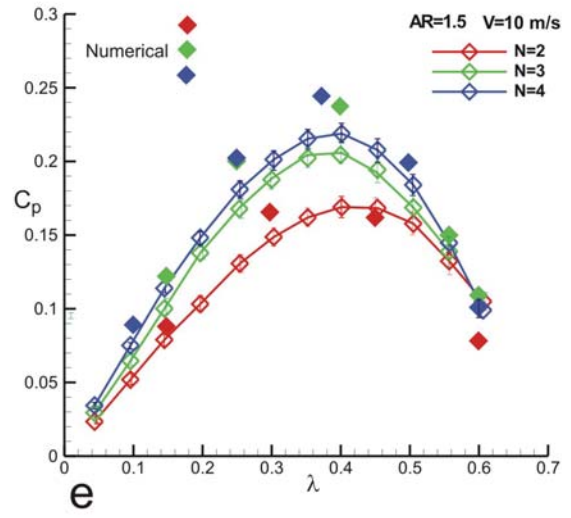
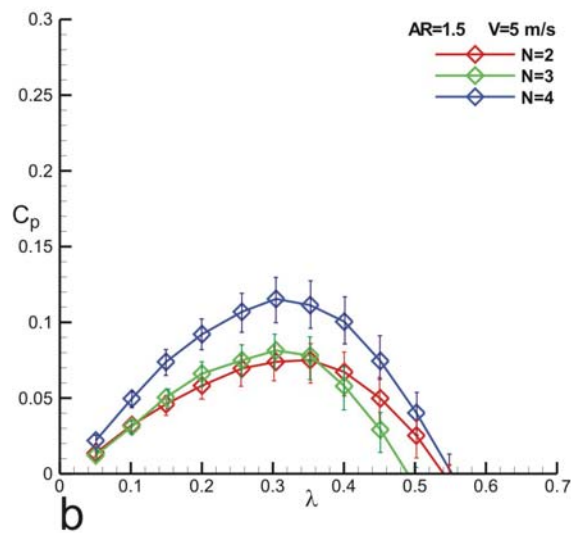
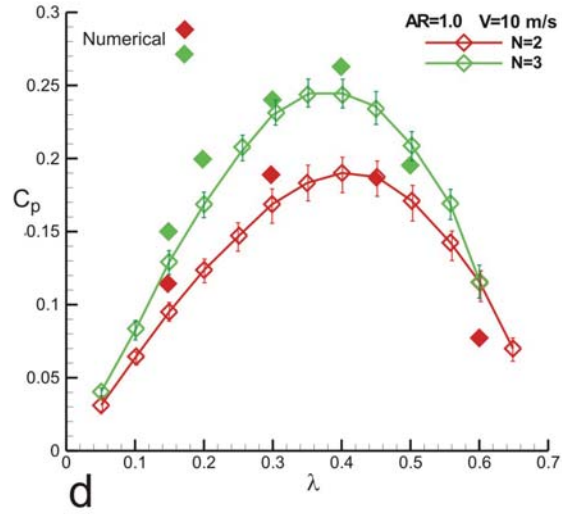
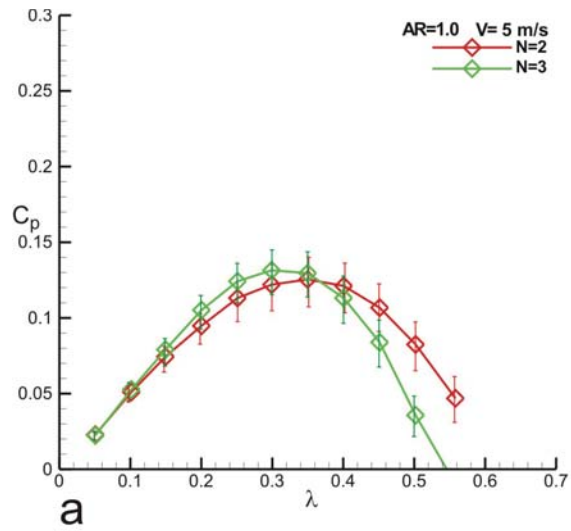


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647 **Figure 4: Effect of mesh density on the accuracy of the predicted power coefficient.**



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650 **Figure 5: Effect of time-step on the accuracy of the predicted power coefficient.**

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Figure 6: Effect of Aspect Ratio, number of blades (Solidity), and wind speed on the average power coefficient of O-VAWT.

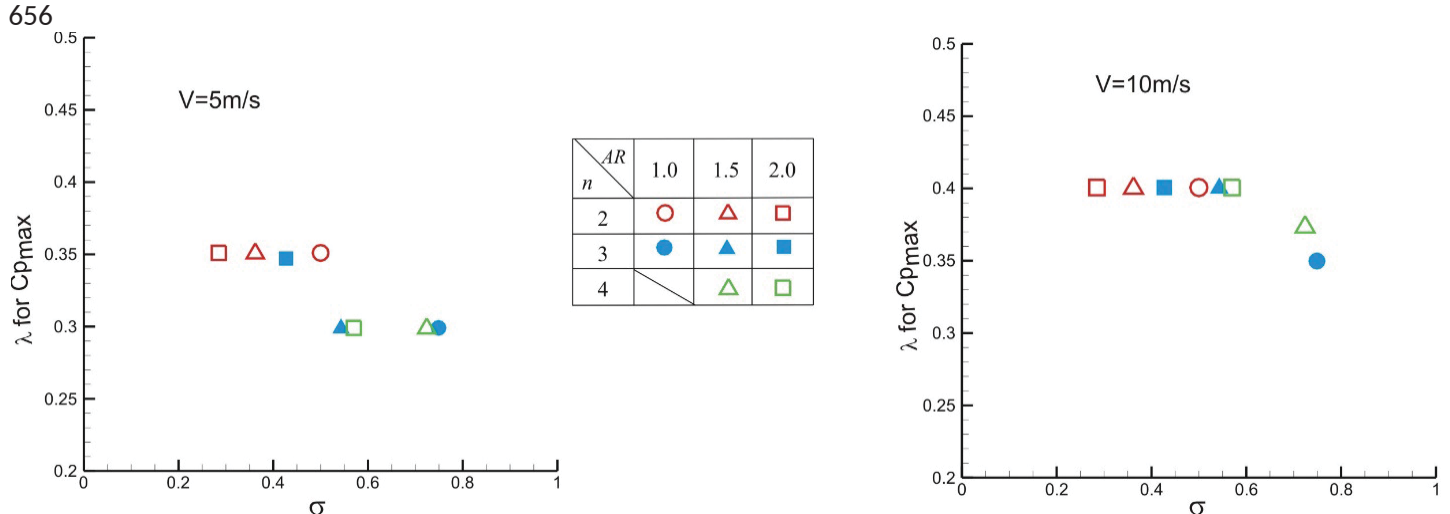


Figure 7: Effect of Aspect Ratio and number of blades (Solidity) on the tip speed ratio corresponding to the maximum power coefficient of O-VAWT at a) $V=5\text{m/s}$, and b) $V=10\text{m/s}$.

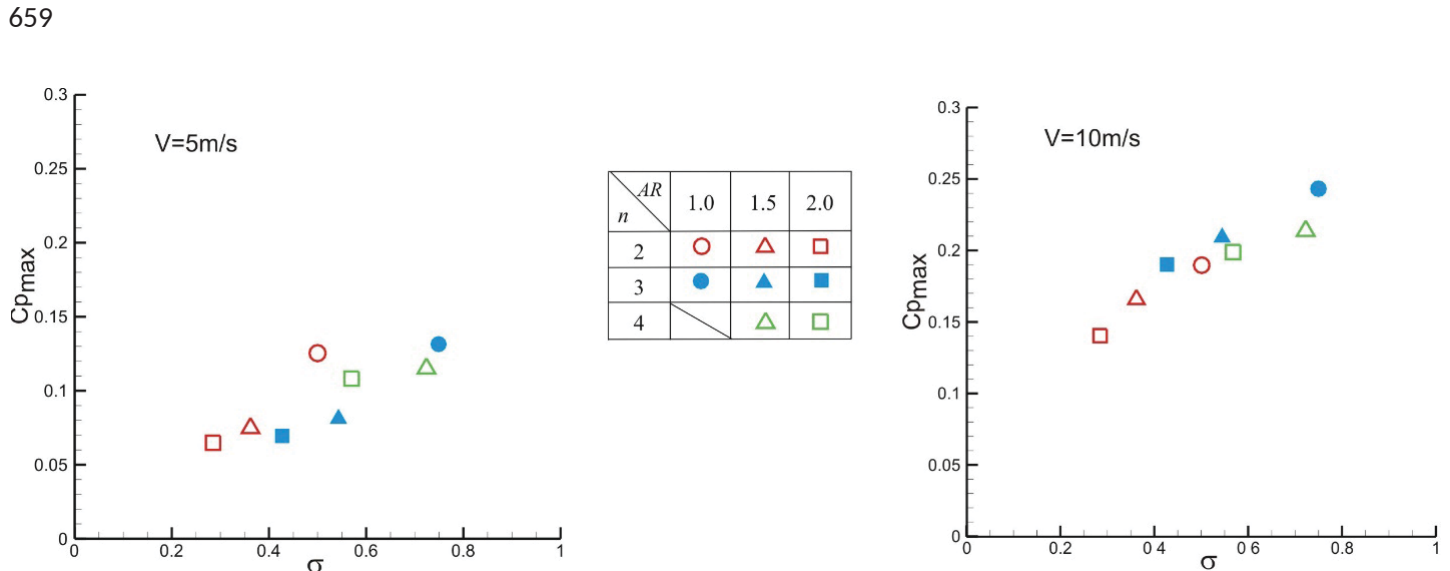
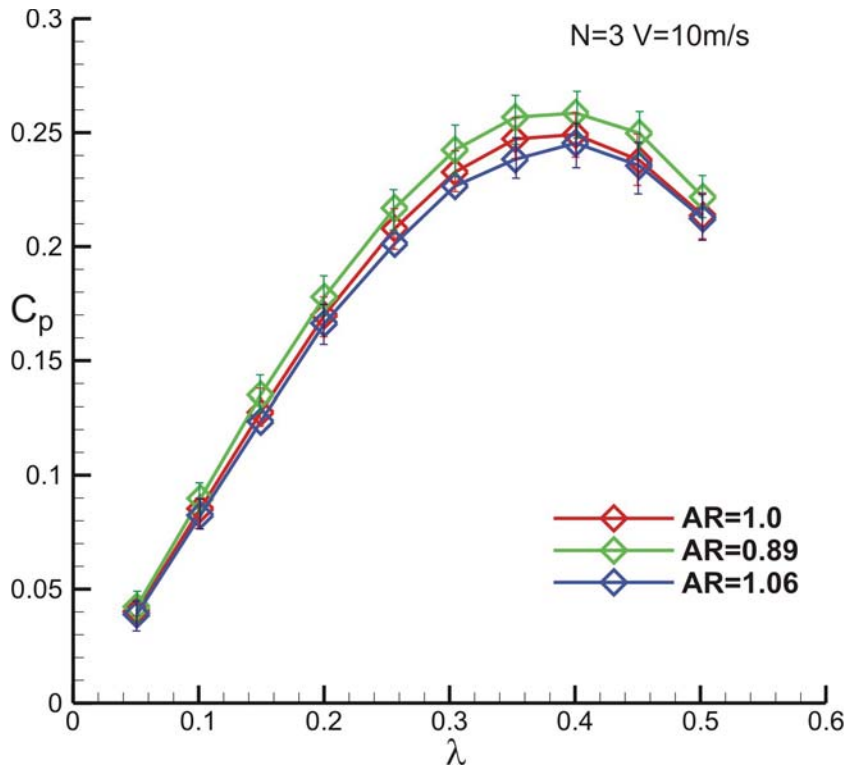


Figure 8: Effect of Aspect Ratio and number of blades (Solidity) on the maximum power coefficient of O-VAWT at a) $V=5\text{m/s}$, and b) $V=10\text{m/s}$.

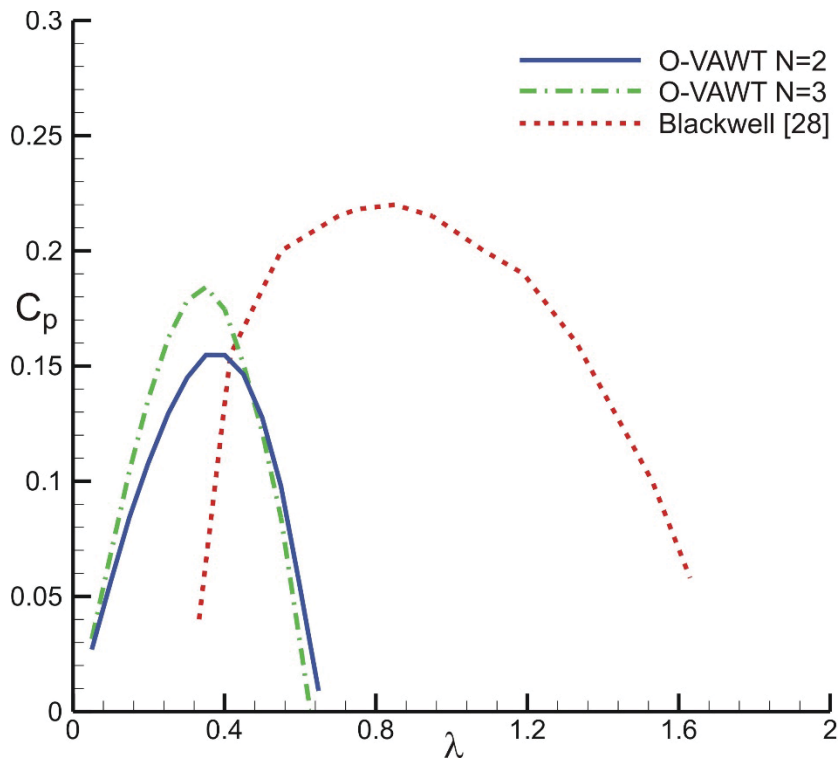
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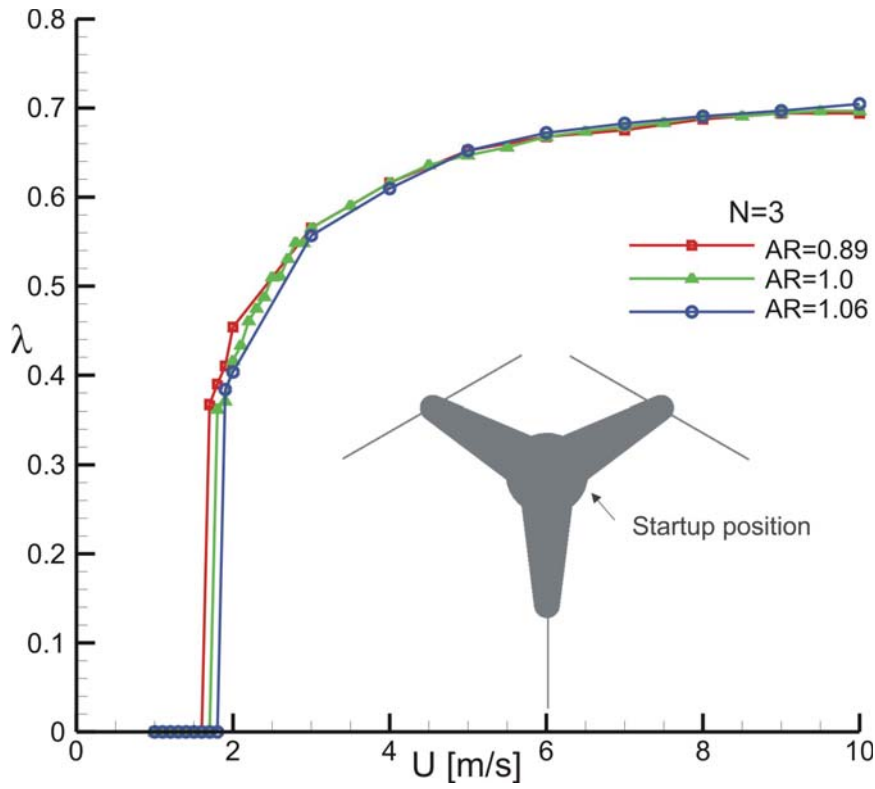
664 **Figure 9: Effect of chord length on the averaged power coefficient of the O-VAWT**

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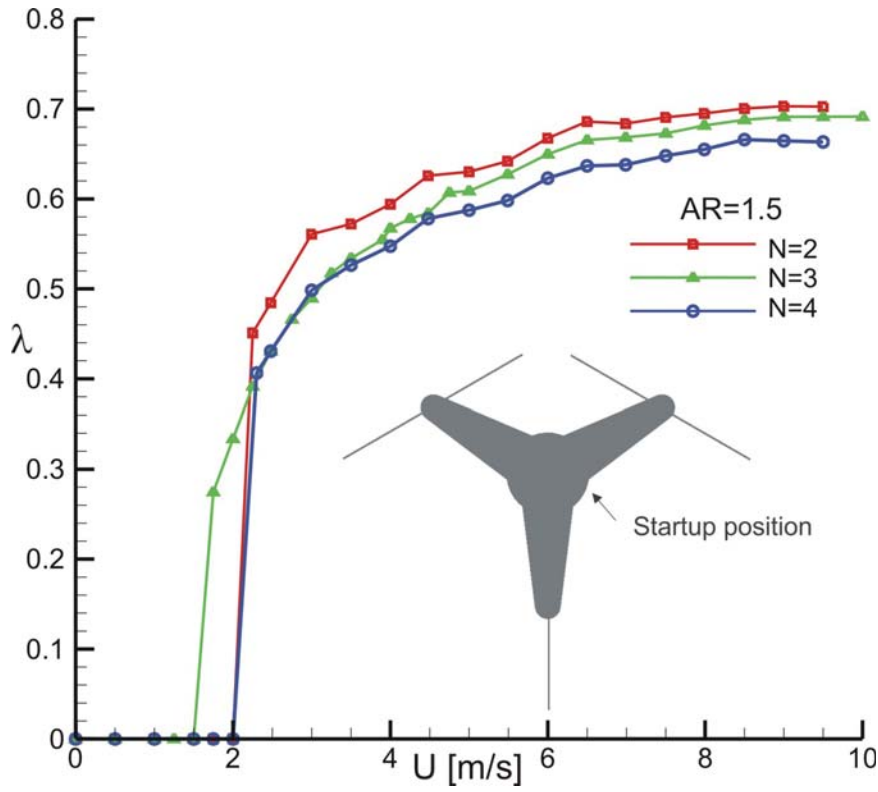
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667 **Figure 10: A comparison of C_p vs. TSR between O-type and S-type VAWTs.**



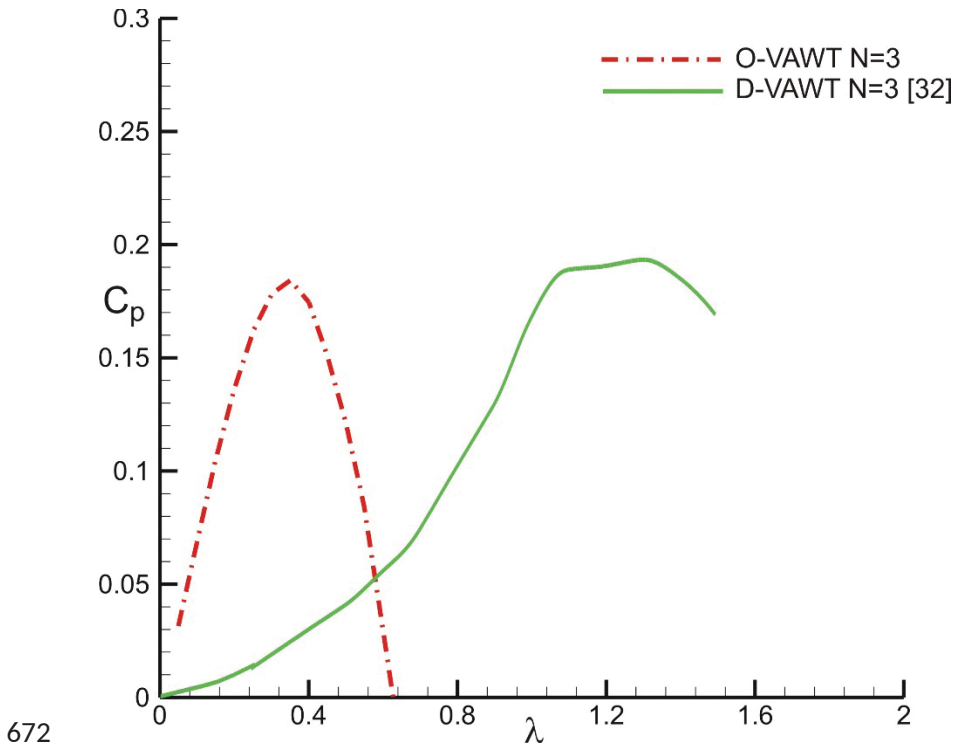
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669 **Figure 11: Effect of chord length on the start-up of the O-VAWT.**



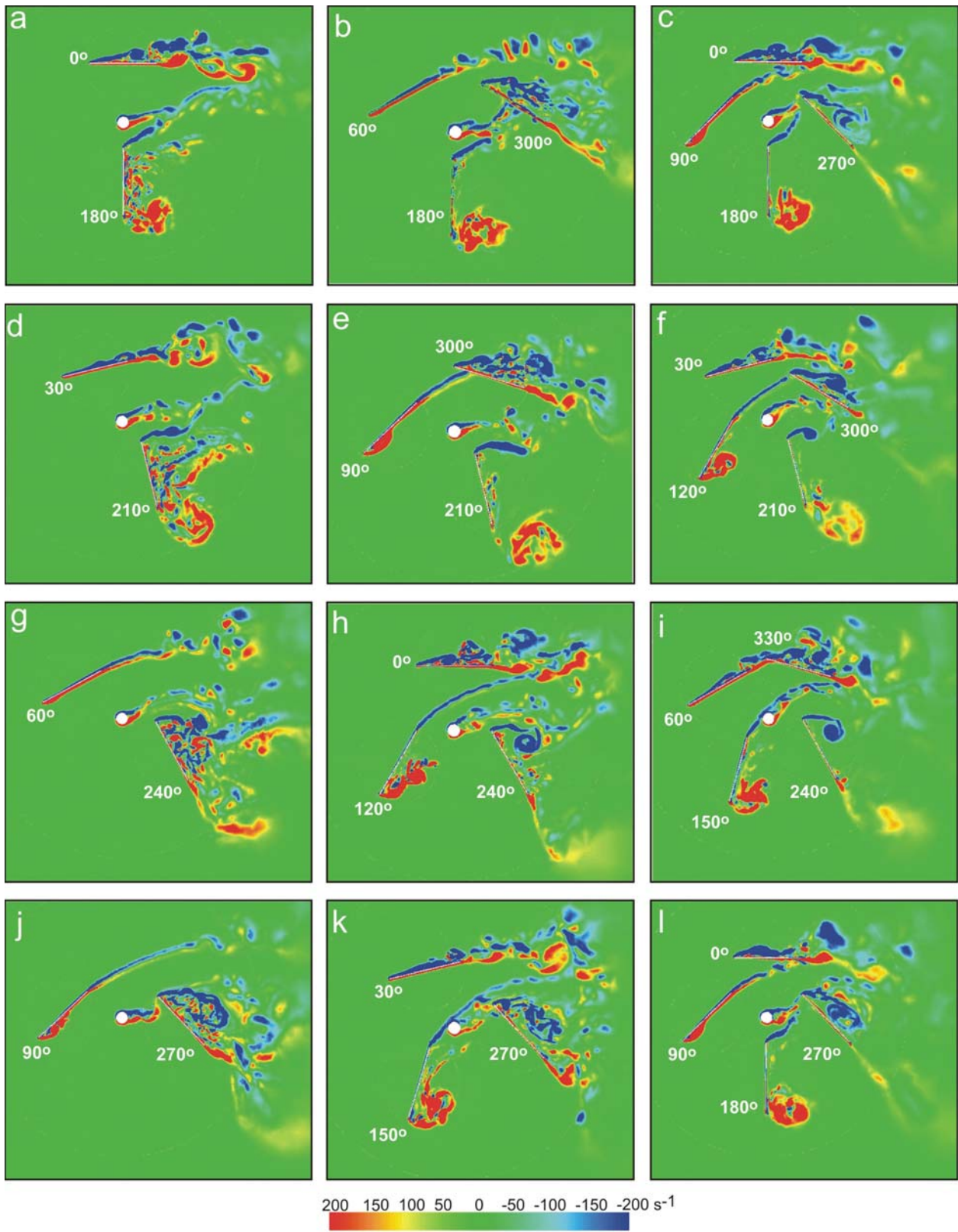
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671 **Figure 12: Effect of number of blades on the start-up of the O-VAWT.**



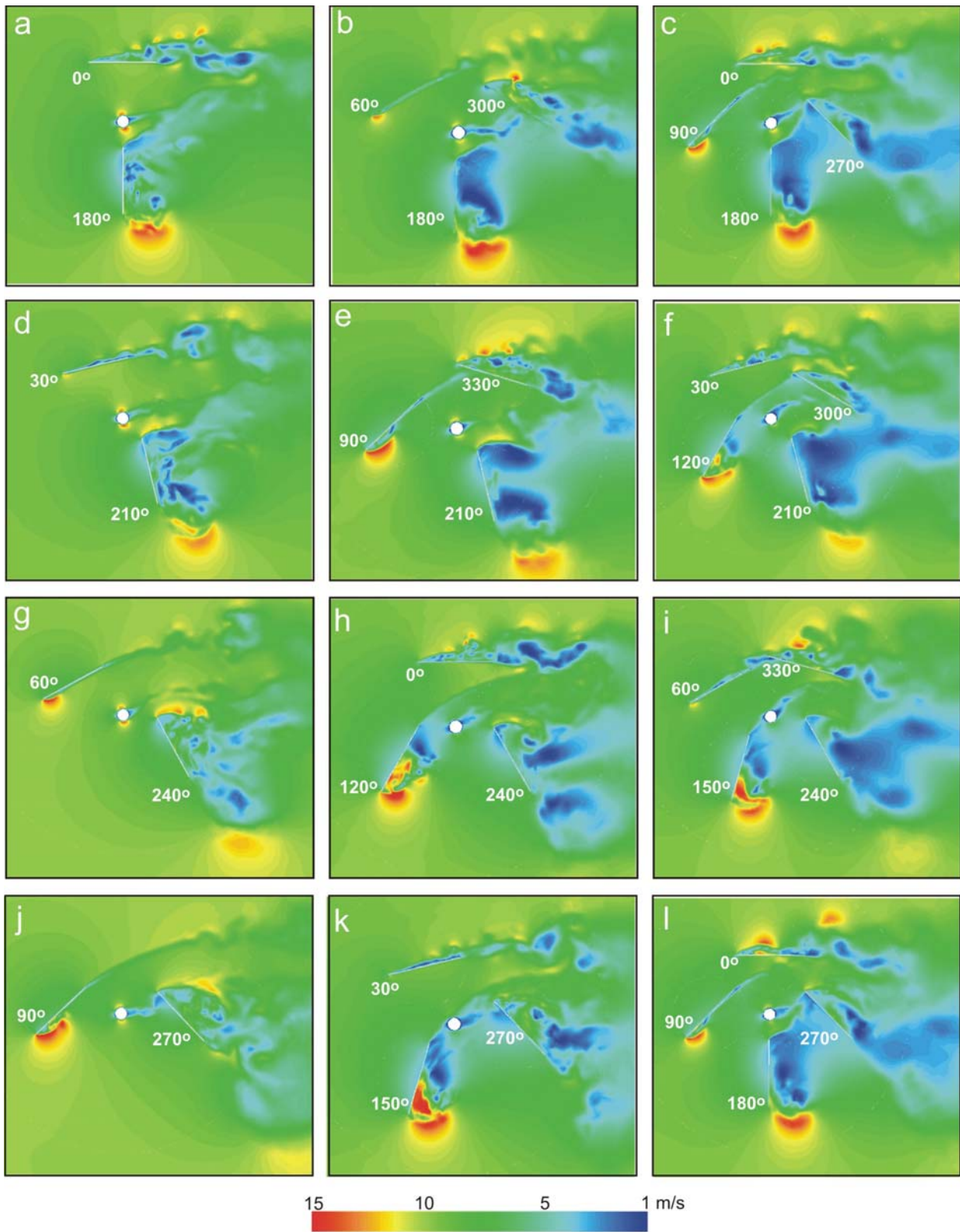
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673 **Figure 13: A comparison of C_p vs. TSR between O-type and D-type VAWTs.**



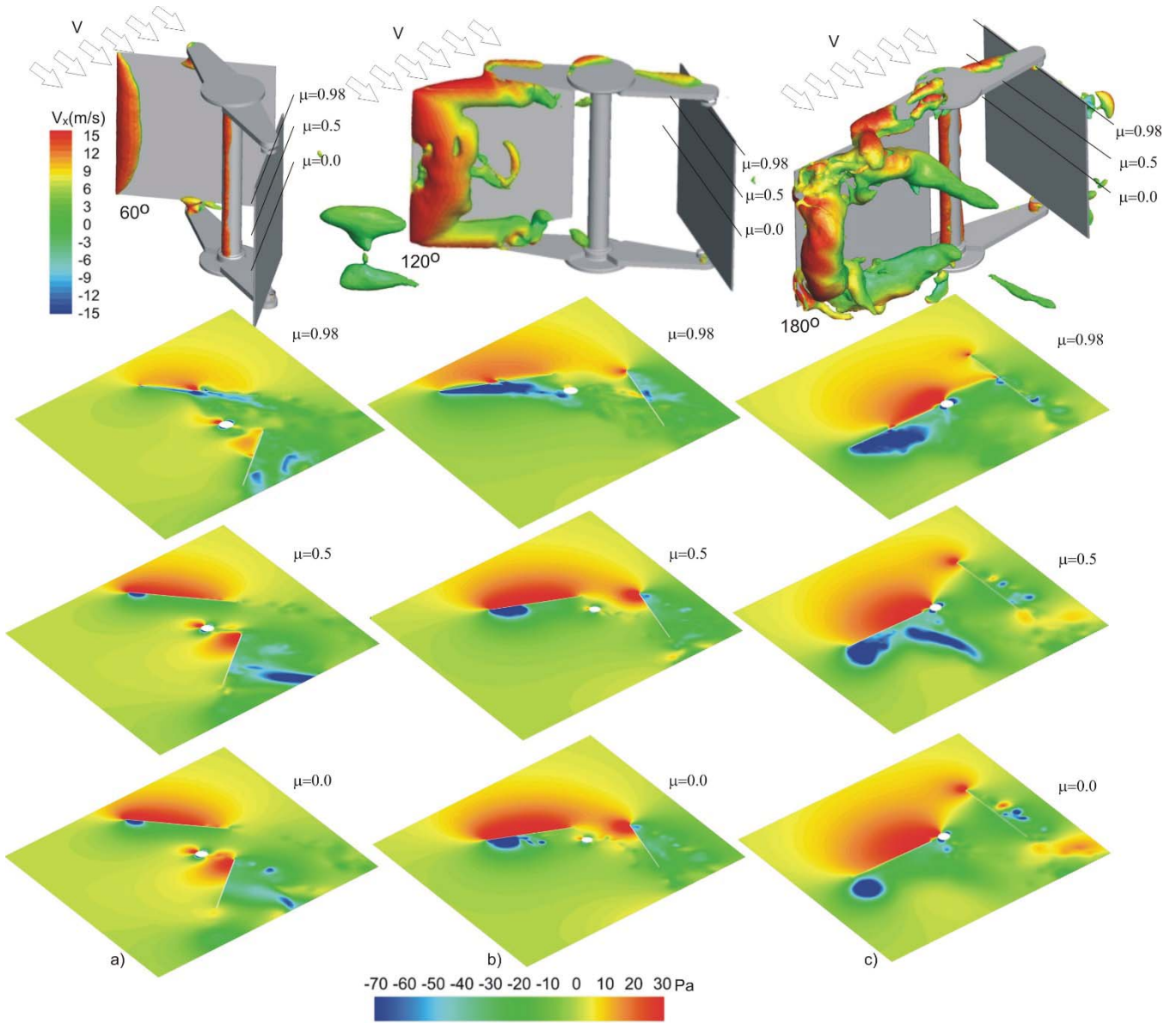
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675 **Figure 14: Instantaneous out-of-plane vorticity field on a mid-plane of O-VAWT at TSR=0.4 and wind**
 676 **speed of 10 m/s for three different rotors**



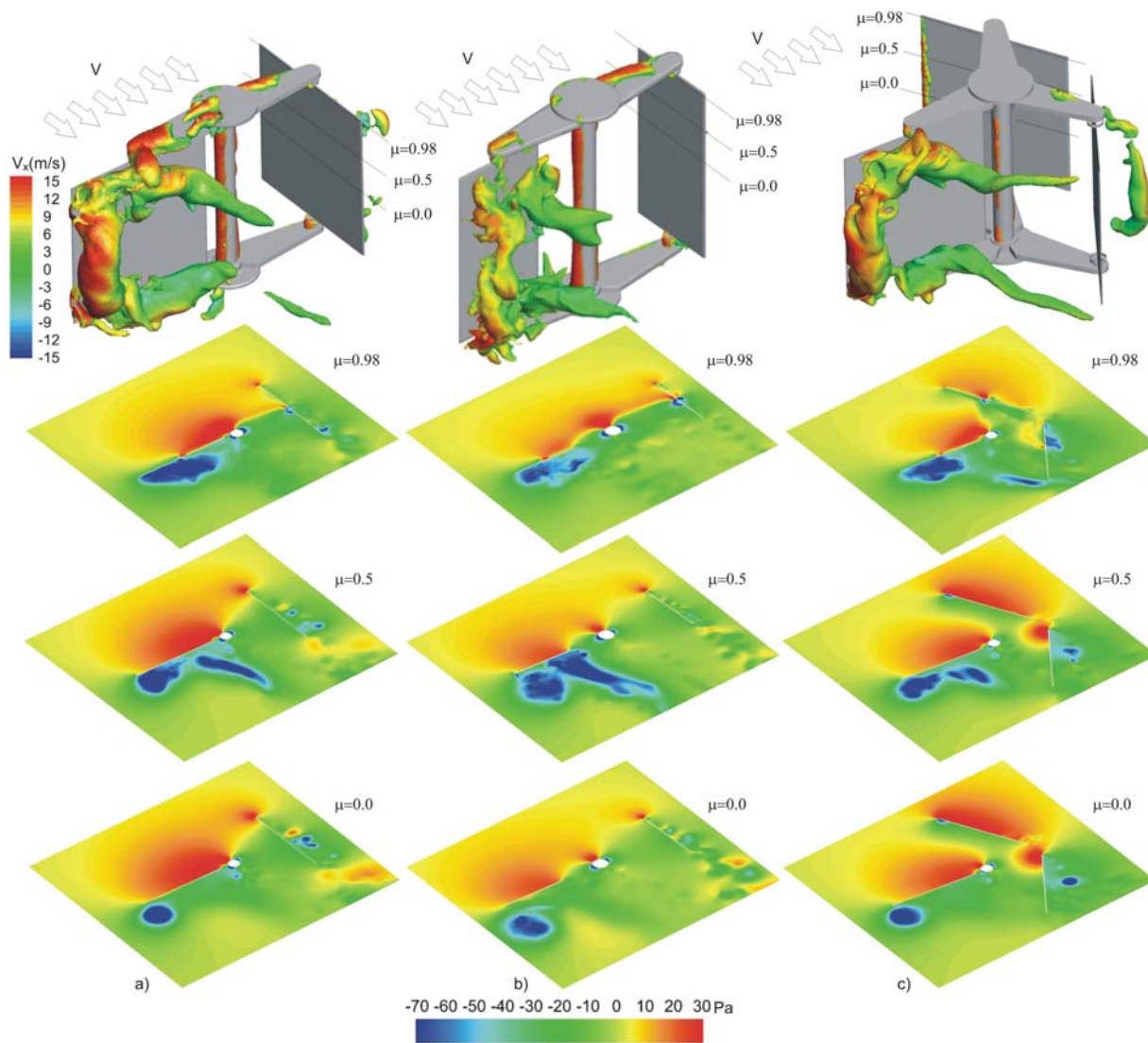
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678 **Figure 15: Instantaneous velocity magnitude field on a mid-plane of O-VAWT at TSR=0.4 and wind speed**
 679 **of 10 m/s for three different rotors.**



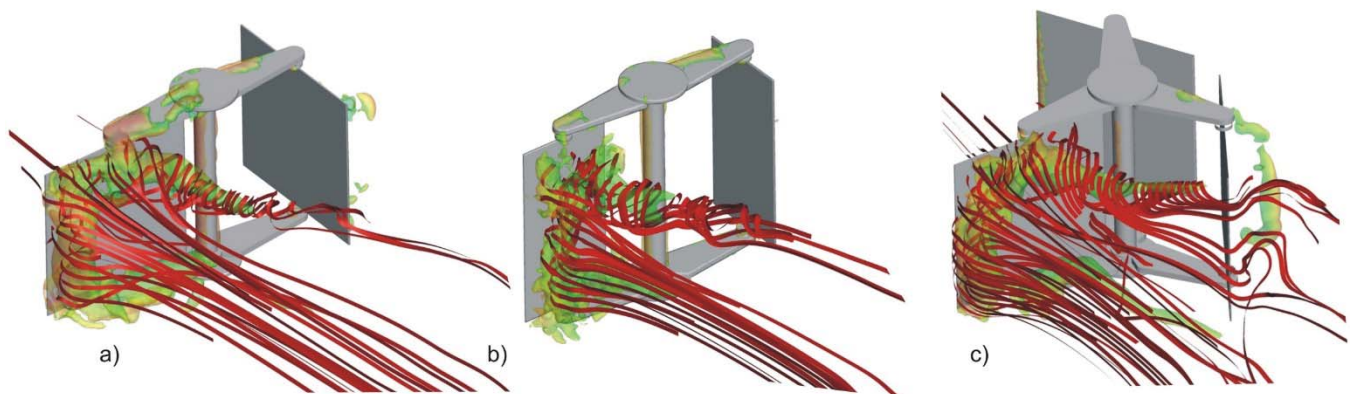
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681 **Figure 16:** Instantaneous pressure Iso-surface of 70 Pa and pressure field on planes at different span-wise
 682 locations of the two-bladed O-VAWT (AR=1) at TSR=0.4 and wind speed of 10 m/s for different blade azimuthal
 683 positions. a) 60° b) 120° c) 180°.



684

685 **Figure 17:** Instantaneous pressure Iso-surface of 70 Pa and pressure field on planes at different span-wise
 686 locations of the O-VAWT at TSR=0.4 and wind speed of 10 m/s. a) N=2, AR=1 b) N=2, AR=2 c) N=3, AR=1.



687

688 **Figure 18:** Streamlines around the 180°-oriented blade of the O-VAWT at TSR=0.4 and wind speed of 10 m/s. a)
 689 N=2, AR=1 b) N=2, AR=2 c) N=3, AR=1.